

39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

### PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. App. 1354(a), 1421 and 1423; 49 U.S.C. 106(g); and 14 CFR 11.89.

#### § 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

**Construcciones Aeronauticas, S.A. (CASA):**  
Docket 92-NM-75-AD.

**Applicability:** All Model C-212-CB, -CC, -CD, -CE, -CF, and -DF series airplanes, certificated in any category.

**Note 1:** This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must use the authority provided in paragraph (e) to request approval from the FAA. This approval may address either no action, if the current configuration eliminates the unsafe condition; or different actions necessary to address the unsafe condition described in this AD. Such a request should include an assessment of the effect of the changed configuration on the unsafe condition addressed by this AD. In no case does the presence of any modification, alteration, or repair remove any airplane from the applicability of this AD.

**Compliance:** Required as indicated, unless accomplished previously. To prevent reduced structural integrity of the airplane, accomplish the following:

(a) For Model C-212-CB series airplanes: Prior to the accumulation of 16,500 total hours time-in-service, or within 6 months after the effective date of this AD, whichever occurs later, replace the horizontal stabilizer to fuselage attach fittings, part numbers 212-31101.05 and 212-31102.05, with part numbers 212-31122.03 and 212-31123.05, respectively, in accordance with the CASA C-212 Aircraft Maintenance Manual, Chapter 5, Section 5-20, task number 55.15.

**Note 2:** Replacement of the attach fittings on Model C-212-CB series airplanes may be accomplished by replacing part numbers 212-31101.05 and 212-31102.05 with part numbers 212-31123.30 and 212-31122.29, respectively.

(b) For all airplanes: Incorporate a revision into the FAA-approved maintenance inspection program that provides for inspection of the Principal Structural Elements (PSE) defined in CASA Supplemental Inspection Document (SID) C-212-PV-01-SID, dated June 1, 1987 (hereinafter referred to as the "Document"), at the later of the times specified in paragraphs (b)(1) and (b)(2) of this AD.

(1) Prior to the accumulation of 20,000 total landings or 20,000 total hours time-in-service, whichever occurs first. Or

(2) Within 9 months after the effective date of this AD.

(c) Any cracked structure detected during the inspections required by paragraph (b) of this AD must be repaired or replaced, prior to further flight, in accordance with the instructions in the Document, or in accordance with other data meeting the certification basis of the airplane that is approved by the FAA or by the Dirección General de Aviación Civil (DGAC).

(d) Within 10 days after accomplishing each inspection required by paragraph (b) of this AD, report the results (positive or negative) of each inspection required by paragraph (b) of this AD to CASA in accordance with the Document. Information collection requirements contained in this regulation have been approved by the Office of Management and Budget (OMB) under the provisions of the Paperwork Reduction Act of 1980 (44 U.S.C. 3501 *et seq.*) and have been assigned OMB Control Number 2120-0056.

(e) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Standardization Branch, ANM-113, FAA, Transport Airplane Directorate. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Standardization Branch, ANM-113.

**Note 3:** Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Standardization Branch, ANM-113.

(f) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

Issued in Renton, Washington, on April 27, 1995.

**Darrell M. Pederson,**

*Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.*

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### 14 CFR Part 39

[Docket No. 95-NM-35-AD]

#### **Airworthiness Directives; Boeing Model 727-100 and -200 Series Airplanes Equipped With an Engine Nose Cowl Installed in Accordance With Supplemental Type Certificate (STC) SA4363NM**

**AGENCY:** Federal Aviation Administration, DOT.

**ACTION:** Notice of proposed rulemaking (NPRM).

**SUMMARY:** This document proposes the adoption of a new airworthiness

directive (AD) that is applicable to certain Boeing Model 727-100 and -200 series airplanes. This proposal would require replacing the attach-nutplates on certain engine nose cowls with washers and self-locking nuts. This proposal is prompted by reports indicating that nose cowls separated (or nearly separated) from the engines of certain airplanes following failure of the engine fan blade and subsequent vibration of the engine, which caused loosening of the attach bolts on the nose cowl of the engine. The actions specified by the proposed AD are intended to prevent the attach bolts from becoming loose, which could result in subsequent separation of the nose cowl from the engine.

**DATES:** Comments must be received by June 12, 1995.

**ADDRESSES:** Submit comments in triplicate to the Federal Aviation Administration (FAA), Transport Airplane Directorate, ANM-103, Attention: Rules Docket No. 95-NM-35-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056. Comments may be inspected at this location between 9 a.m. and 3 p.m., Monday through Friday, except Federal holidays.

The service information referenced in the proposed rule may be obtained from VALSAN Partnership Ltd., Aviation Products Management, Product Support Office, 39450 Third Street East, suite 121, Palmdale, California 93550. This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington.

**FOR FURTHER INFORMATION CONTACT:** Phil Forde, Aerospace Engineer, Airframe Branch, ANM-120S, FAA, Transport Airplane Directorate, Seattle Aircraft Certification Office, 1601 Lind Avenue SW., Renton, Washington 98055-4056; telephone (206) 227-2771; fax (206) 227-1181.

#### **SUPPLEMENTARY INFORMATION:**

##### **Comments Invited**

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications shall identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this notice may be changed in light of the comments received.